

Relative Distance Control of Uncooperative Tethered Debris

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January 16, 2023

Outline

- 1 Introduction
- 2 System Modeling and Control
- 3 Simulation Results
- 4 Sensitivity Study
- 5 Conclusion
- 6 Estimation and Control

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Tether-Based Active Debris Removal

Active debris removal by means of interconnection through a flexible tether is a promising option. The primary means of capture through such a tether are:

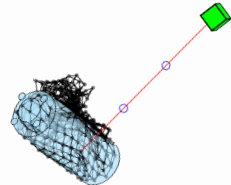
- Nets
- Harpoons

The flexible connection introduces **complex dynamics in the post-capture phases** of the ADR mission that the chaser craft must manage, particularly the slackness in the tether.

Relative distance controllers have been proposed (Jaworski et al., 2017; Cleary and O'Connor, 2016).

Objectives:

- Propose PD and PID controllers
- Investigate dynamics of the tethered system



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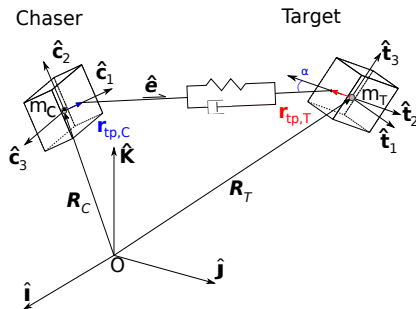
Model Assumptions and Layout

Assumptions

- 1 Rigid chaser (cube) and target (rectangular prism)
- 2 No external forces other than gravity and tether tension
- 3 Kelvin-Voigt tether model

Important frame definitions

- $\mathcal{O} = [\hat{I}, \hat{J}, \hat{K}]$
- $\mathcal{C} = [\hat{c}_1, \hat{c}_2, \hat{c}_3]$
- $\mathcal{T} = [\hat{t}_1, \hat{t}_2, \hat{t}_3]$



α = alignment angle

Kinematics and Dynamics

Translational dynamics: $m\ddot{\mathbf{R}} = -m\mu \frac{\mathbf{R}}{||\mathbf{R}||^3} + \mathbf{T} + \mathbf{U}$

$\mathbf{T} \rightarrow$ Tension $\mathbf{U} \rightarrow$ Control Input

Target attitude: represented by a unit quaternion $\mathbf{q} = [q_0, \mathbf{q}_v]^T$, with $\mathbf{q}_v = [q_1, q_2, q_3]$.

Kinematic relationship between the attitude and angular rates:

$$\dot{\mathbf{q}} = \frac{1}{2} \begin{bmatrix} -q_1 & -q_2 & -q_3 \\ q_0 & -q_3 & q_2 \\ q_3 & q_0 & -q_1 \\ -q_2 & q_1 & q_0 \end{bmatrix} \begin{bmatrix} \omega_x \\ \omega_y \\ \omega_z \end{bmatrix}$$

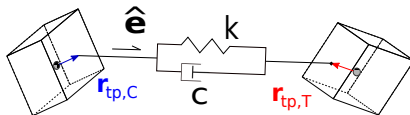
Rotational dynamics for body X :

$$\dot{\boldsymbol{\omega}} = J_X^{-1}({}^X \mathbf{r}_{tp,X} \times {}^X A^O \mathbf{T} - \boldsymbol{\omega} \times J_X \boldsymbol{\omega} + \boldsymbol{\tau})$$

where \mathcal{X} is its respective body fixed frame (e.g. ${}^C \mathbf{r}_{tp,C}$).

Tether Modeling

- Cannot support compression
- No twisting or bending modeled



Tether Tension Vector:
$$\mathbf{T} = \begin{cases} T \hat{\mathbf{e}} & \text{if } (l > l_0) \wedge (T > 0), \\ 0 & \text{if } (l \leq l_0) \vee (T \leq 0). \end{cases} \quad T = k(l - l_0) + c\dot{l}$$

k = stiffness c = damping l_0 = natural length δl = elongation

Tether Heading Vector:
$$\hat{\mathbf{e}} = \frac{\mathbf{R}_T + {}^O \mathbf{r}_{tp,T} - \mathbf{R}_C - {}^O \mathbf{r}_{tp,C}}{\|\mathbf{R}_T + {}^O \mathbf{r}_{tp,T} - \mathbf{R}_C - {}^O \mathbf{r}_{tp,C}\|}$$

Tether Length and Length Rate:
$$l = \|\mathbf{R}_T + {}^O \mathbf{r}_{tp,T} - \mathbf{R}_C - {}^O \mathbf{r}_{tp,C}\|$$

$$\dot{l} = (\mathbf{V}_T + {}^O \mathbf{A}^T \boldsymbol{\omega}_T \times^T \mathbf{r}_{tp,T} - \mathbf{V}_C - {}^O \mathbf{A}^C \boldsymbol{\omega}_C \times^C \mathbf{r}_{tp,C}) \cdot \hat{\mathbf{e}}$$

Relative Distance Control

Process Variable: $e = \Delta l + l_0 - l$

Δl = desired elongation

PID/PD Input Thrust Magnitudes:

- $F_{PID} = K_P e + K_I \int_0^t e dt + K_D \dot{e}$

- $F_{PD} = K_p e + K_d \dot{e}$

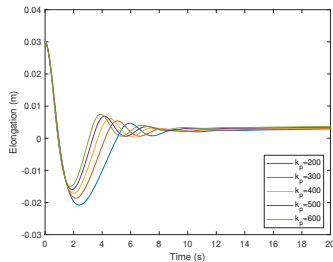
PID/PD Input Thrust Directions:

- $\mathbf{F}_{PID} = -F_{PID} \hat{\mathbf{e}}$

- $\mathbf{F}_{PD} = -F_{PD} \hat{\mathbf{e}}$

For comparison, open loop control:

$$\mathbf{F}_{OL} = -20 \hat{\mathbf{V}}_C$$



Heuristic tuning of gains, with desired elongation of 0.01 m.

Chaser Attitude Control

Desired chaser axes:

$${}^{\mathcal{O}}A_d^C = [\hat{\mathbf{e}} \mid (\hat{\mathbf{e}} \times \hat{\mathbf{R}}_C) \times \hat{\mathbf{e}} \mid \hat{\mathbf{e}} \times \hat{\mathbf{R}}_C]$$

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Simulations for Controller Performance Analysis

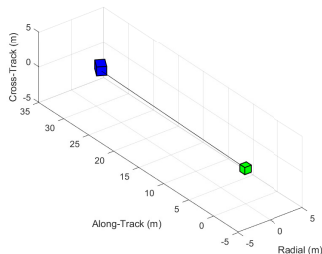
Two scenarios:

- Initially taut tether: $\delta l = 3 \times 10^{-5}$ m
- Initially slack tether: $\delta l = -1$ m

Initial $\alpha = \pi/6$ rad: imperfect capture scenario.

Scenarios simulated for 500 s.

$$\begin{aligned}
 K_P &= 300 \text{ N/m} & K_D &= 2000 \text{ Ns/m} \\
 K_I &= 300 \text{ N/ms}
 \end{aligned}$$

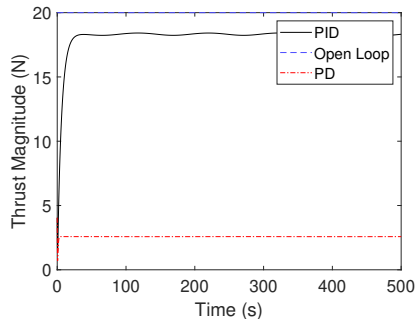
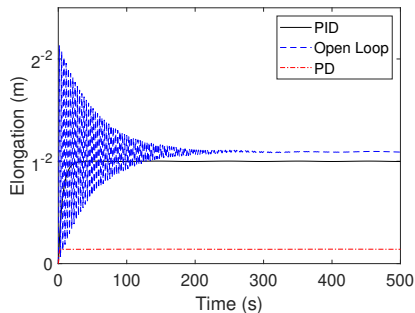


System parameters for the simulation scenarios.

Parameter	Value	Parameter	Value
J_C (kg-m ²)	diag(83.3, 83.3, 83.3)	m_C (kg)	500
J_T (kg-m ²)	diag(15000, 3000, 15000)	m_T (kg)	3000
${}^C r_{tp,C}$ (m)	$[0.5, 0, 0]^T$	k (N/m)	1573
${}^T r_{tp,T}$ (m)	$[0, -0.875, 0]^T$	c (Ns/m)	16
l_0 (m)	30	ω_T (rad/s)	$[0, 0.05, 0]^T$

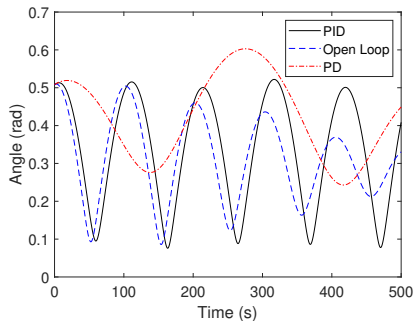
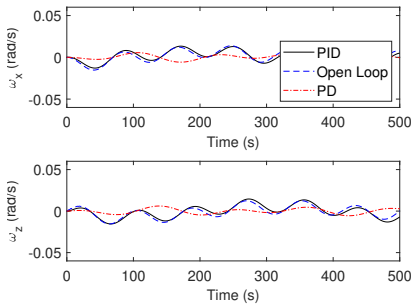
Initially Taut Tether

- PID control achieves δl of 0.01 m
- PD control steady state δl at ~ 0.0014 m.



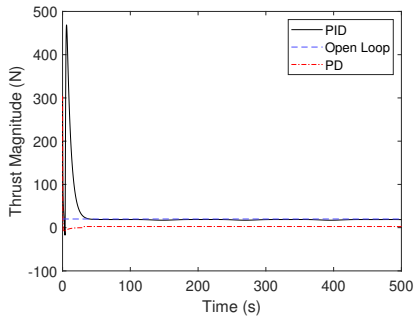
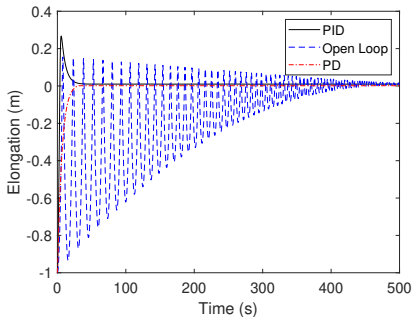
Initially Taut Tether - Continued

- Lower applied moment from PD control.
- Open loop α behavior result of its thrust direction.



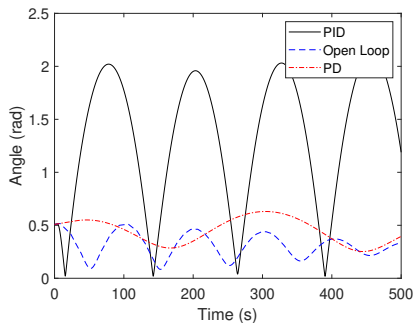
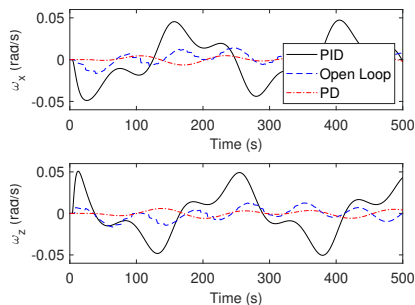
Initially Slack Tether

- PID control results in 0.25m peak δl
- PD control steady state δl again ~ 0.0014 m



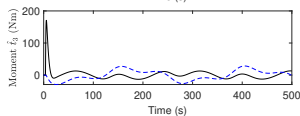
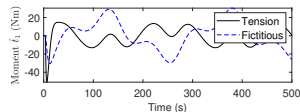
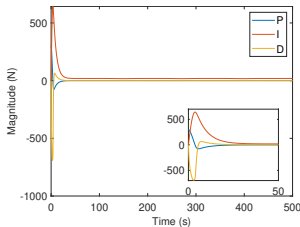
Initially Slack Tether - Continued

- Relaxation of tether tension in PID control causes large α .
- PD and open loop result in similar ω and α as the previous scenario.



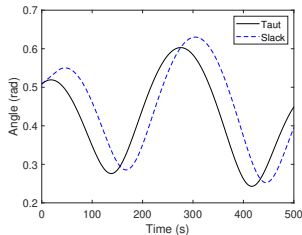
Initially Slack Tether - Continued

- PD α difference caused by time when $T > 0$.
- For PID, K_I responsible for large peak δI .
- For PID, steady-state applied moment insufficient to counter initial angular impulse.



Discussion

- PID controller introduces dangerous motion of the debris; T larger than w/ PD control.
- PD control does not achieve desired ΔI ; similar behavior in both simulations.



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Sensitivity Study Setup

System parameters (target's especially) are uncertain. Sensitivity study conducted on **key variables**:

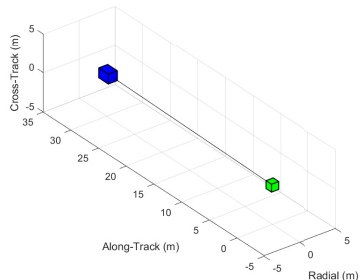
- Target Mass
- Target Moments of Inertia
- Initial Angular Rates
- Initial Relative Distance

Outputs to analyze the variations in the dynamics:

- Peak alignment angle α_{max}
- Control effort E_c
- Integrated squared norm of the angular rates W

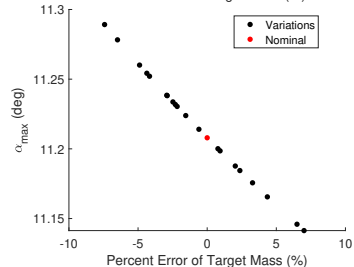
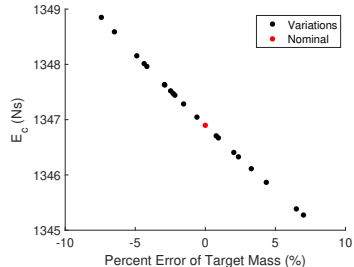
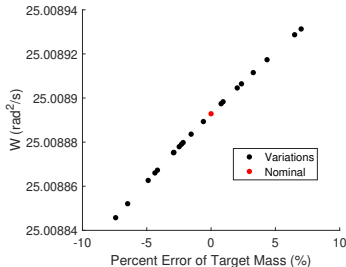
$$W = \int_0^t \|\omega\|^2 dt \quad E_c = \int_0^t |F_{PD}| dt$$

Variable	Error Bound
Target Mass	± 300 kg
Target Inertia Matrix	$\pm \text{diag}[3000 \ 600 \ 3000]$ kg-m ²
Target Angular Rates	± 0.04 rad/s
Initial Rel. Distance	± 0.3 m



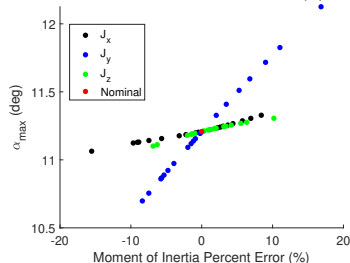
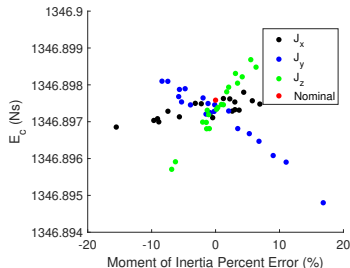
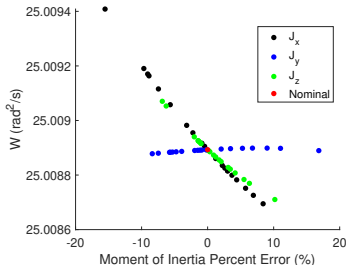
Target Mass Study

- Decreased mass \rightarrow more responsive target.
- Angular rate integral W increases with mass.
- Differences marginal.



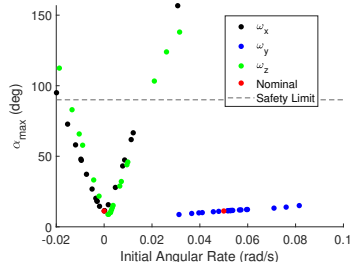
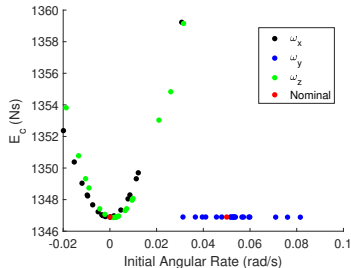
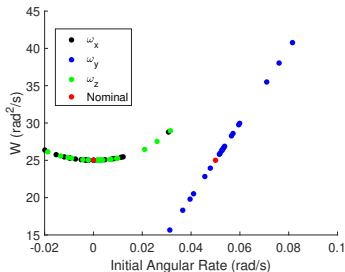
Target Moments of Inertia Study

- Asymmetry in J_x and $J_z \rightarrow$ non-zero $\dot{\omega}_y$.
- More significant variations in α_{max} and W .
- E_c differences order of 10^{-3} Ns.
- Varying J_y has the largest impact on α_{max} .



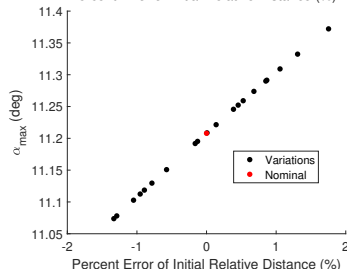
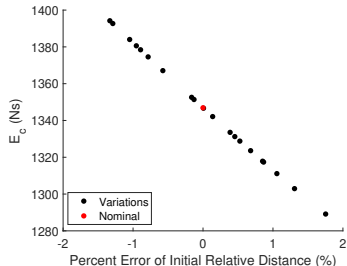
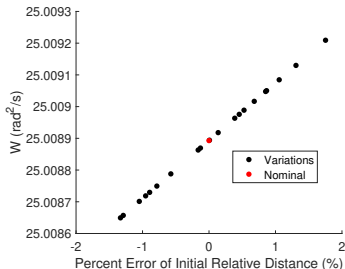
Target Angular Rates Study

- Quadratic trends in ω_x and ω_z .
- Tumbling motion exceeds ability of PD controller.
- Inertial velocity of target attachment point increased.



Initial Relative Distance Study

- E_c increases as chaser gets closer.
- Effects on α_{max} and W caused by chaser's motion when $T < 0$.
- PD control capable of handling this uncertainty.



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Conclusion

Proposed PD and PID controllers to maintain a relative distance between the chaser and target.

- **PID control:**

- Possible tether winding → Unsuitable for this purpose.

- **PD control:**

- Induced more relaxed angular motion on the target.
 - Capable of handling uncertainty in target inertia properties and initial relative distance.
 - Cannot maintain safe target attitude motion in certain circumstances.

Future Work

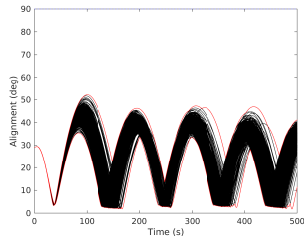
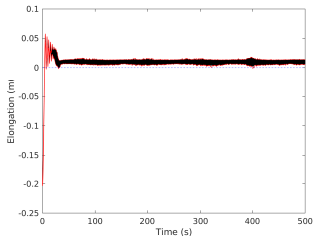
- Apply PD control alongside target state estimation.
- Active target attitude control.

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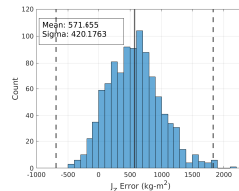
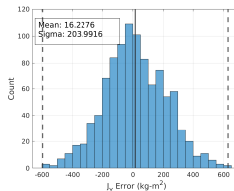
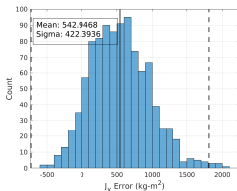
Online Estimation and Control

- Currently assuming image processing and subsequent update is achieved instantaneously at t_k (i.e., at t_k , \mathbf{X}_k^- is updated to \mathbf{X}_k^+ which is used for control calculation).
- Measurement frequency of 10 Hz. and tracking 5 feature points.
- Using PID control with saturation at 50 N to get best estimates at the moment.



Online Estimation and Control

- The final estimate distributions for J_x , J_y , and J_z are within 3.62%, 0.541%, and 3.81% of the true values, respectively.



Thank you for your time.
Questions?

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Acknowledgements: Funded by the National Science Foundation
Award# CRII: RI:2105011